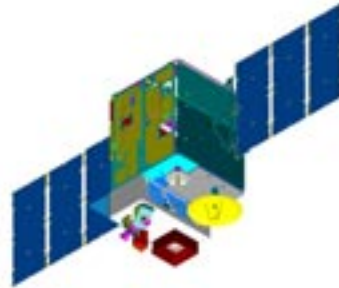


**Geosynchronous Imaging Fourier Transform Spectrometer -Indian Ocean METOC Imager
(GIFTS-IOMI)**

Pioneering Geo-Science and Operations for the New Millennium





GIFTS-IOMI Mission

Briefing to ESE Technology Strategy Team

November 29, 2001

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GIFTS-IOMI Project Manager

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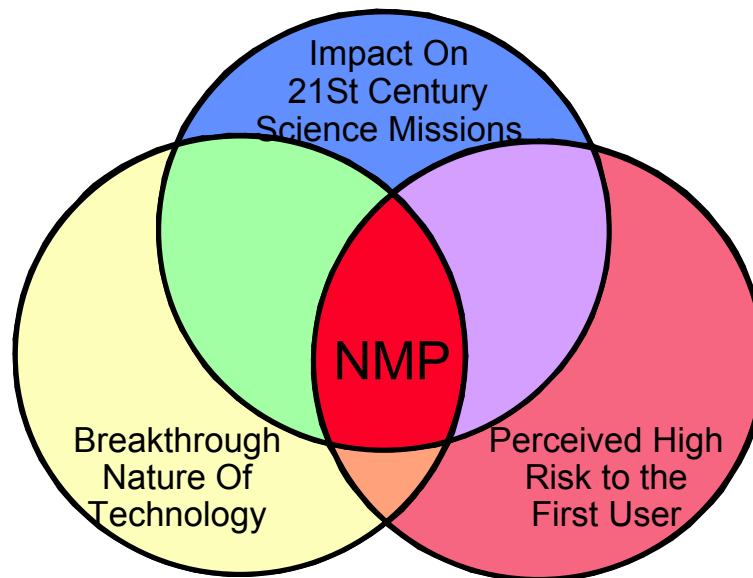
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New Millennium Program



Flight Validation of Breakthrough Technologies to
Benefit Future Earth Science Missions



Breakthrough technologies

- Enable new capabilities to meet Earth Science needs
- Reduce costs of future missions

Flight validation

- Mitigates risks to first users
- Enables rapid technology infusion into future missions



NMP Program Objectives

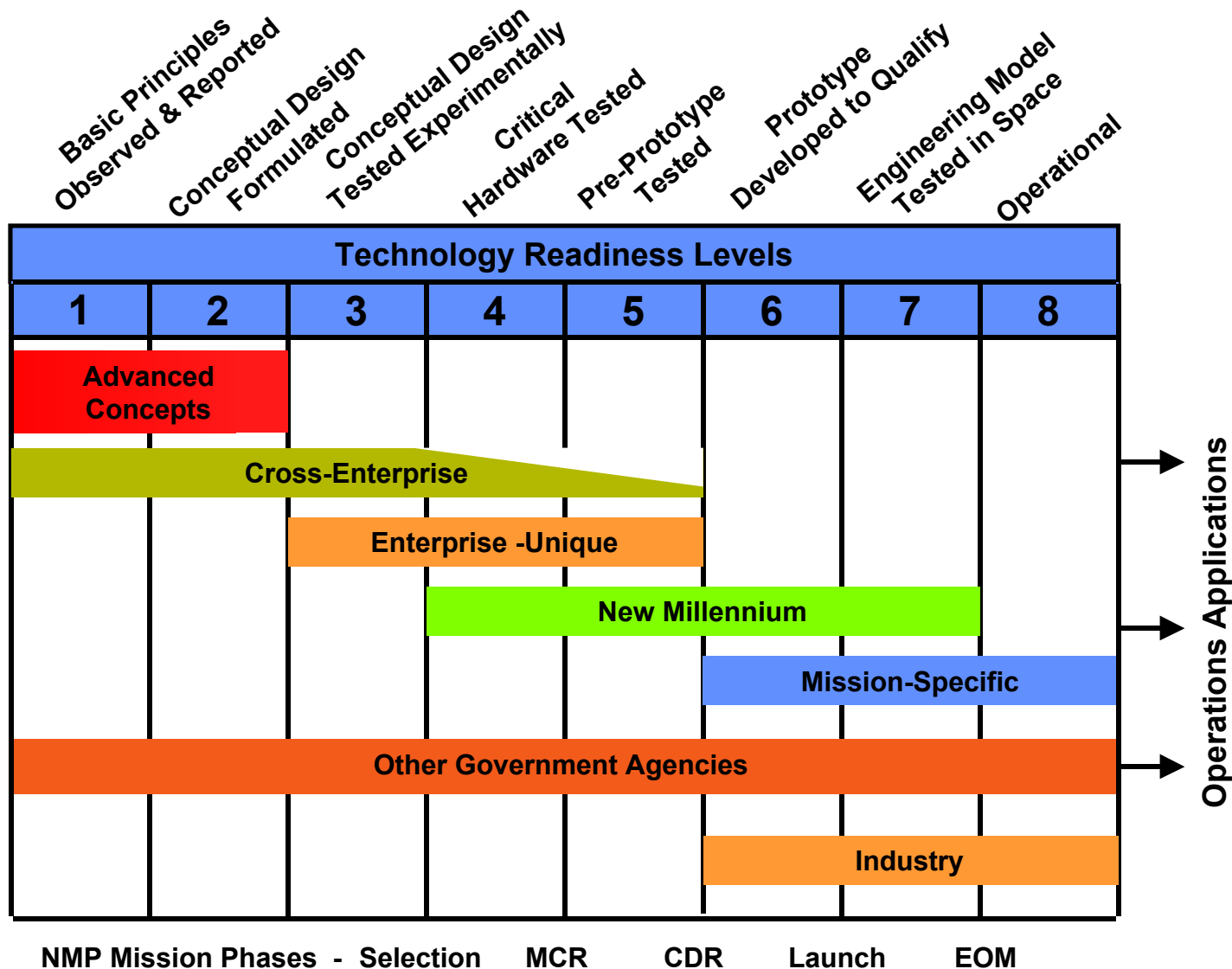


- Both the Space Science and Earth Science Enterprises utilize the NMP as a primary entity to flight validate key emerging technologies to achieve NASA's strategic objectives with exciting 21st century earth and space science missions. To fulfill this critical role, the NMP will have three primary objectives:
 - Identify and select technologies for flight validations that optimize the benefits to the Space and Earth Science programs;
 - Develop and implement effective validation flight projects as appropriate testbeds to mitigate the risks for using the selected technologies; and
 - Facilitate the infusion of the validated technologies into future science mission opportunities.

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Role of NMP in Technology Development



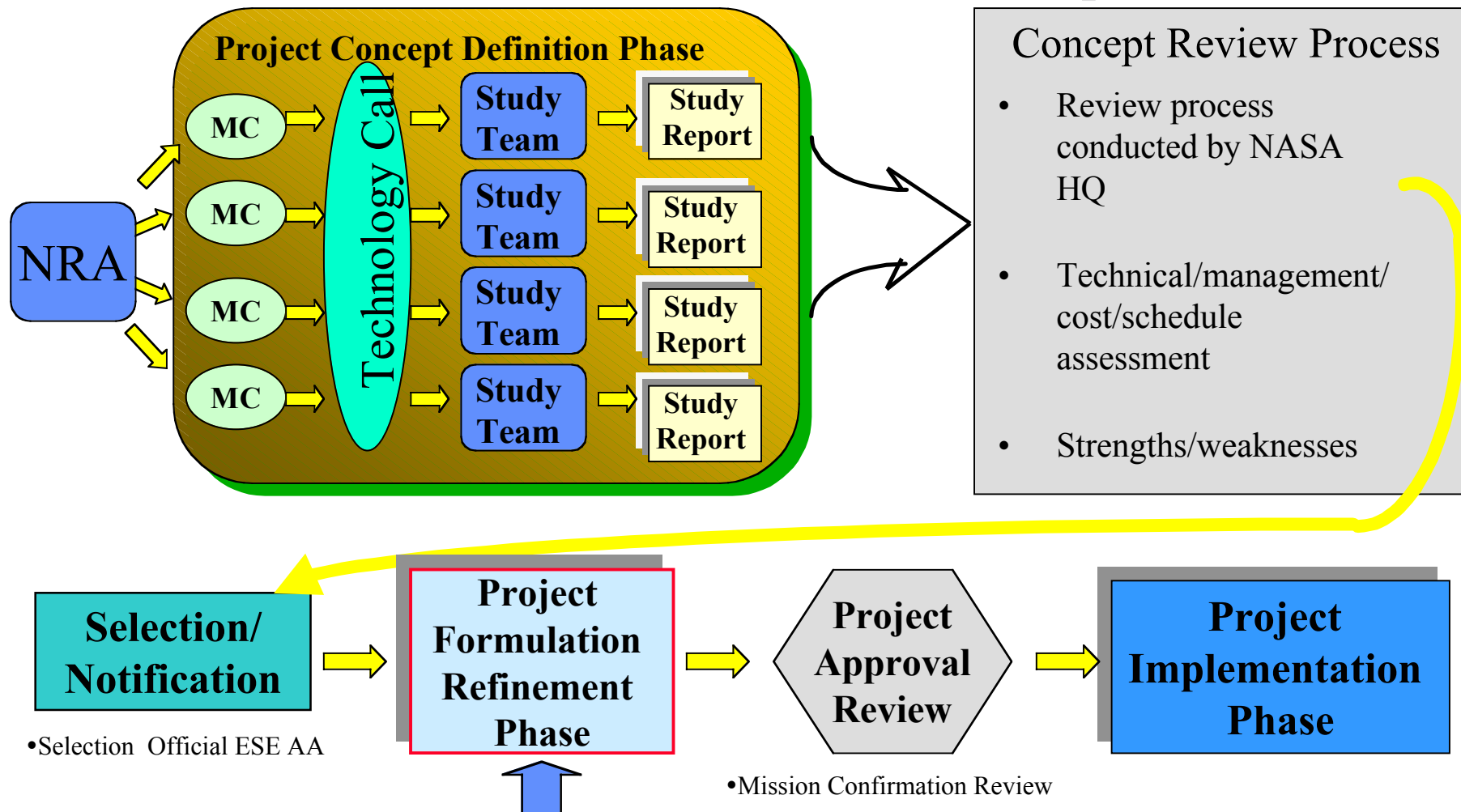


New Millennium Program Project Formulation



Code Y Process

- Focused on measurement concepts

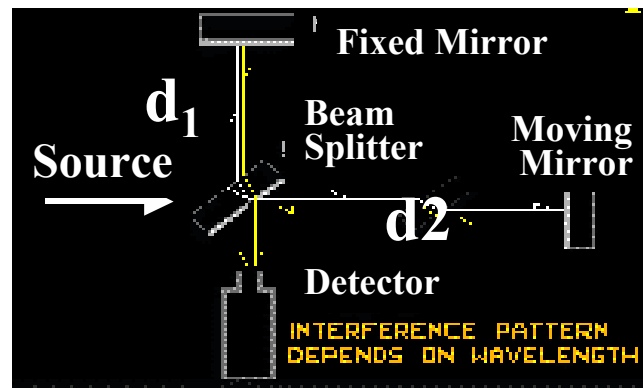


• Adopted for EO3

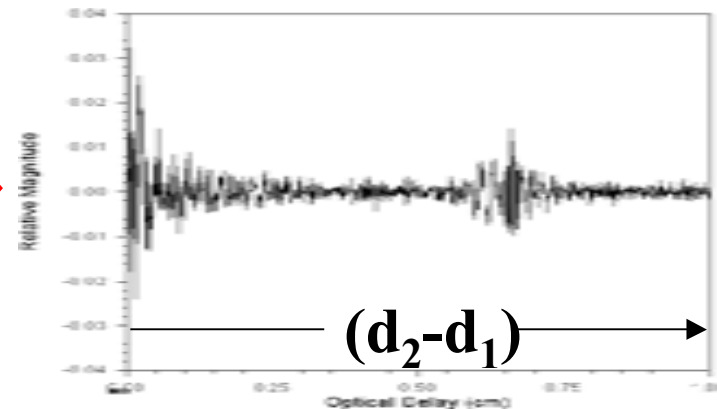
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High Spectral Resolution Imagery

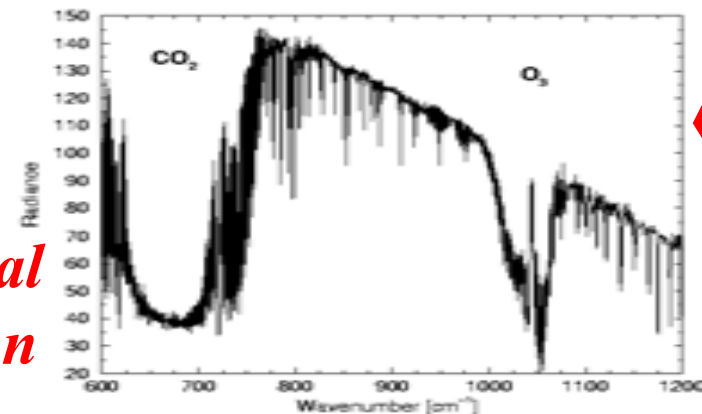
Michelson Interferometer (FTS)



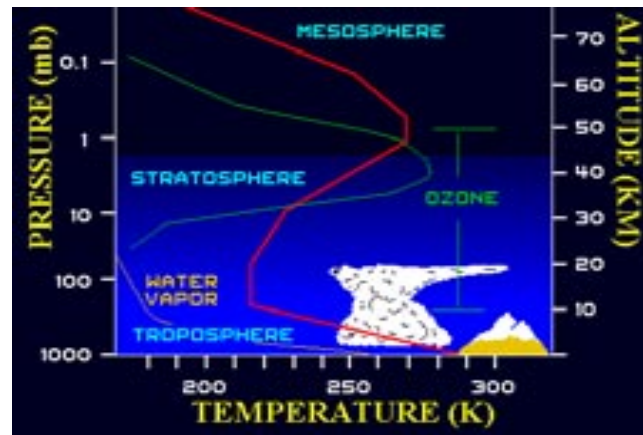
Interferogram



Fourier Transformation



Numerical Inversion



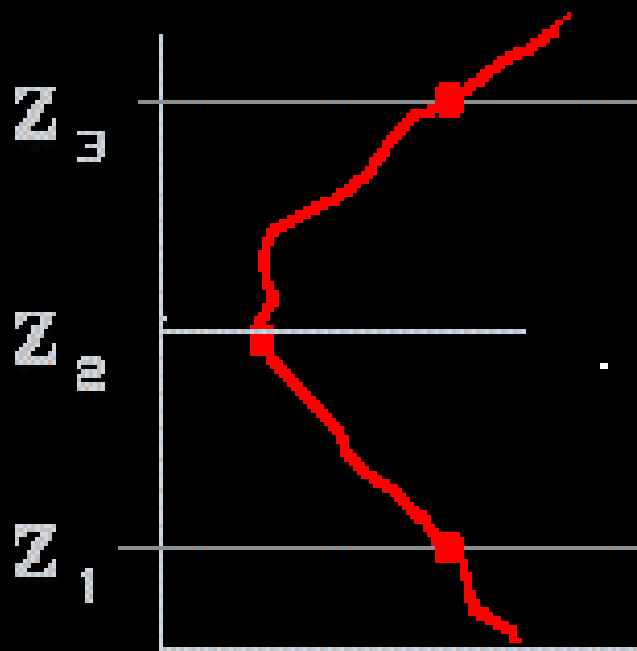
Vertical Sounding

Radiance Spectrum

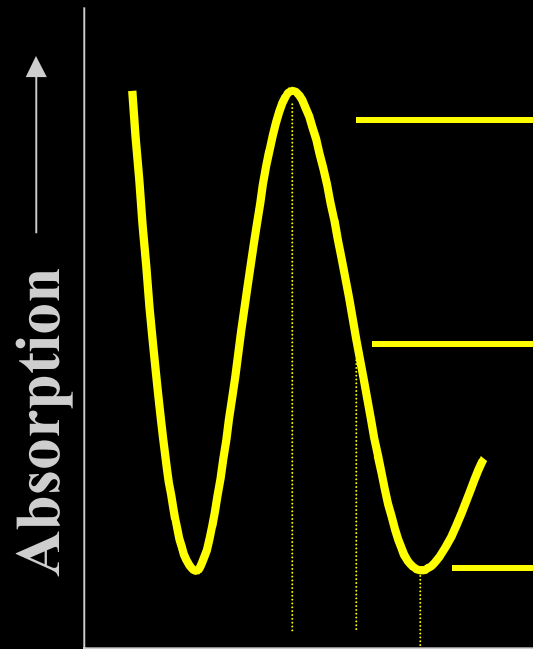
Fourier Transform Spectrometer (FTS) Provides Simultaneous Measurement of the Vertical Dimension

Vertical Sounding Technique

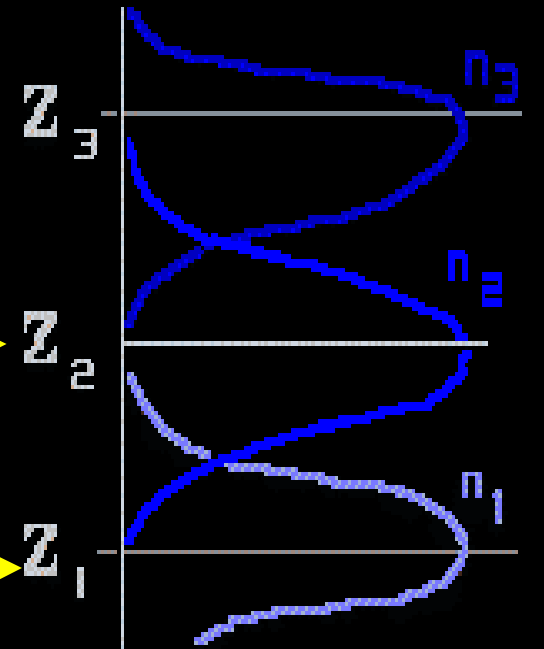
Wavelength Converts to Altitude



$T(z)$

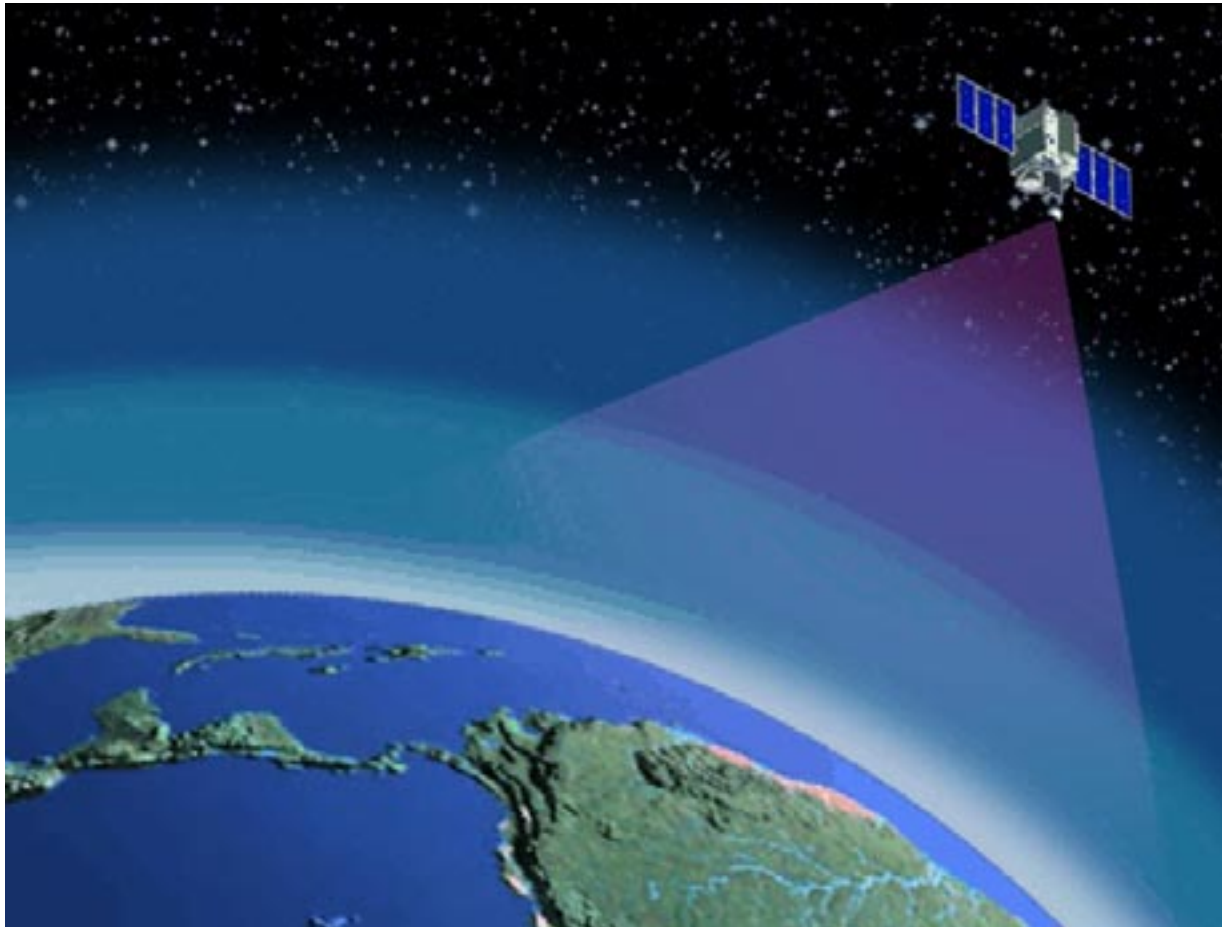


n_3 n_2 n_1
Wavelength



Energy
Contribution

GIFTS Sampling Characteristics



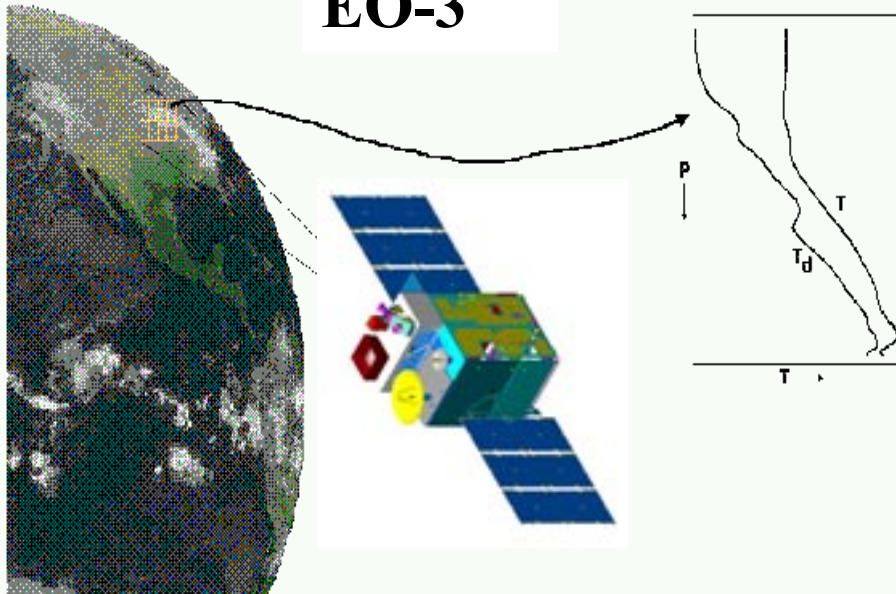
- Two 128x 128 Infrared focal plane detector arrays with 4 km footprint size
- One 512 x 512 Visible focal plane detector array with 1 km footprint size
- Field of Regard
512 km x 512 km at satellite sub-point
- Ten second full spectral resolution integration time per Field of Regard

GIFTS Upper Air Observation Density



for Geosynchronous Satellite Application

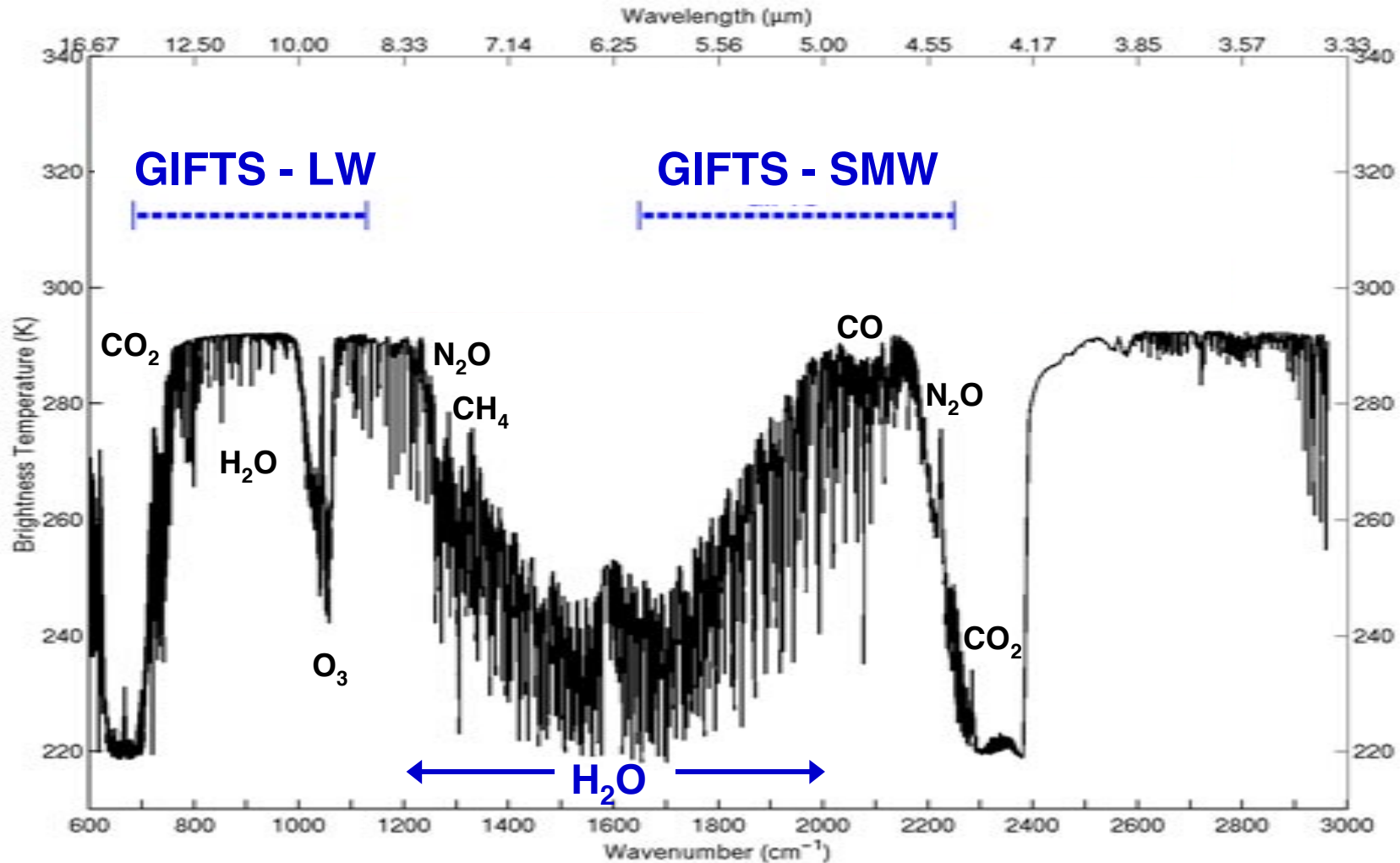
EO-3



**Global Conventional
Balloon Systems
provide ~1000/day
widely separated
vertical soundings over
land areas**

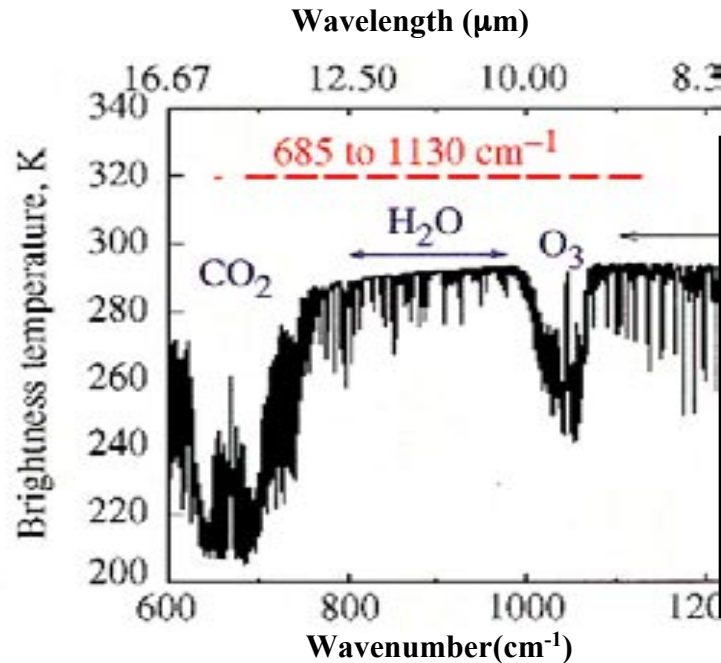
**GIFTS will provide ~
100,000/min closely
spaced vertical
temperature &
moisture soundings
over land and sea!**

Spectral Coverage



Spectral Coverage is Optimized for Multi Spectral Imagery Applications and Temperature and Water Vapor Sounding

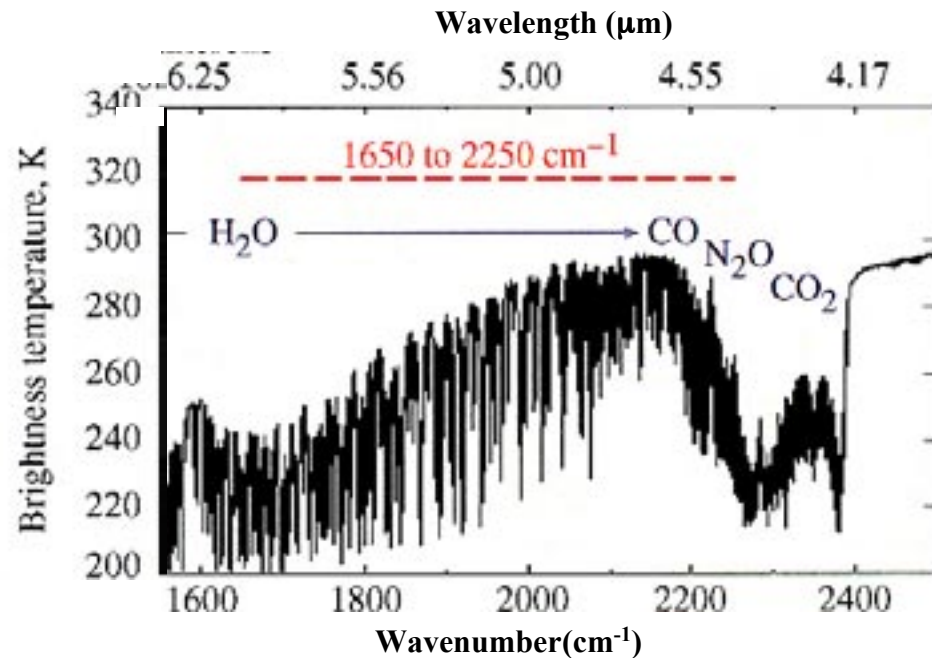
IR Measurement Objective



Longwave IR Focal Plane:

(8.8 to 14.6 microns)

- **Temperature sounding**
- Ozone profiling
- Cloud altitude
- Surface temperature
- “Invisible” cirrus cloud detection
- Dust/Aerosol optical depth



Shortwave IR Focal Plane:

(4.4 to 6.1 microns)

- **Water vapor sounding**
- Carbon Monoxide profiling
- Boundary layer temperature sounding
- Surface temperature
- Stratus cloud detection



GIFTS Measurement Objectives

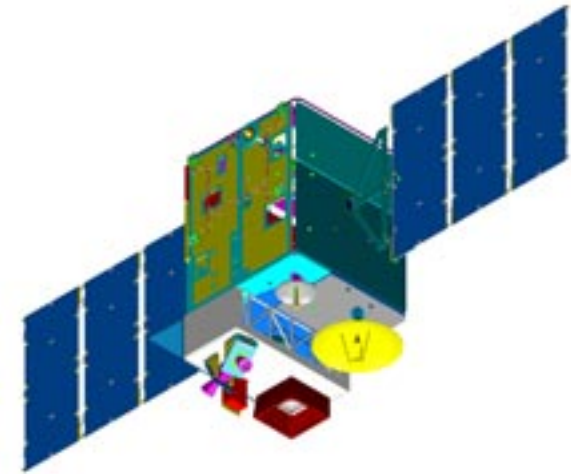


Key Technologies

- *Fourier Transform Spectrometer*
- *Large area format focal plane array*
- *Advanced data readout and digital signal processing electronics*
- *Redundant low mass low power cryogenic coolers*

Operational Constraints

- *Launch date: Fall 2005 to geosynchronous orbit*
- *Operational life: 1 yr NMP/NOAA, 5 yr Navy*
- *Resources: Mass <130 kg, Power <420 w, Data Rate <65 Mbit/sec*



Measurement Capability

- *Measure temperature, water vapor, tracer wind, and chemical composition distribution with high spatial and temporal resolution to enable revolutionary improvements in operational weather observation, prediction and air quality monitoring.*
- *Provide dynamic observations of cloud spectral radiance and associated atmospheric properties.*
- *Observe the transport of radiatively active pollutants and greenhouse gases.*



GIFTS Technology Validation Objectives



TO1. Imaging Interferometer

- Demonstrate a cryogenic Michelson interferometer optimized for sounding applications.

TO2. LFPA and Advanced Cryogenic Cooling

- Demonstrate large area detector arrays and readouts and associated miniaturized cryocoolers.

TO3. High Speed Signal Processing

- Demonstrate high-speed, ultra-low-power signal processing.

TO4. Data Compression

- Demonstrate on-board real-time signal processing and data compression.

TO5. Autonomous Pointing and Control

- Demonstrate autonomous pointing and control systems for precise image stabilization and feature tracking.

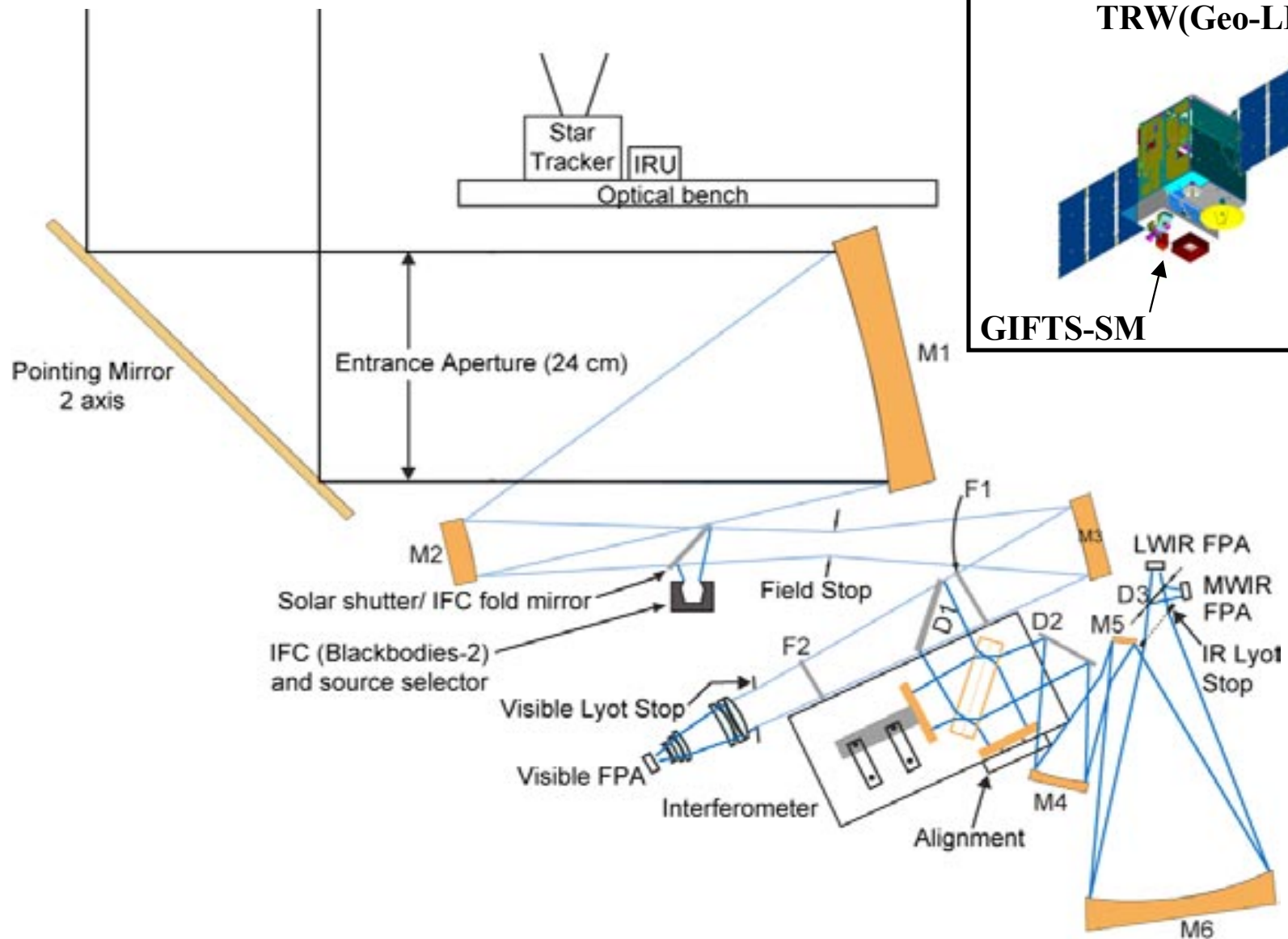
TO6. Low Power Radiation Tolerant Microelectronics

- Demonstrate radiation protection and ultra-low-power electronics.

TO7. Lightweight Structures and Optics

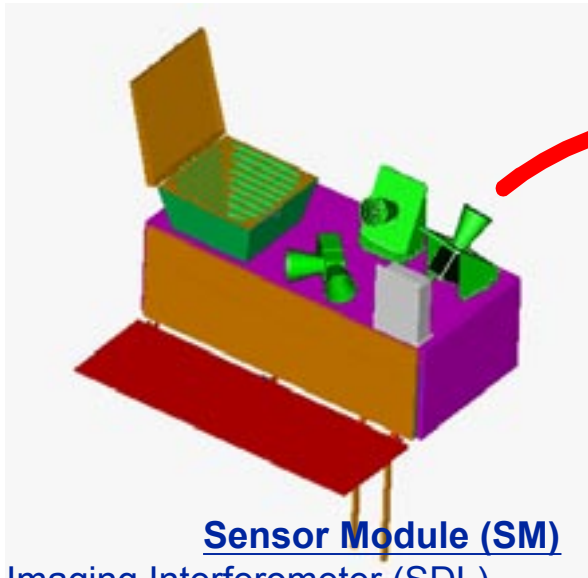
- Demonstrate lightweight optics and structures to minimize instrument mass.

GIFTS Sensor Module Design



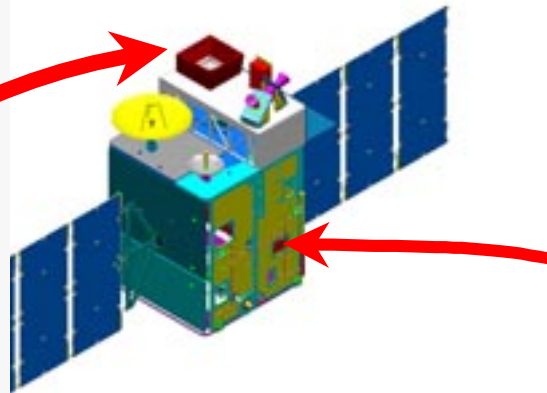


GIFTS Technologies



Sensor Module (SM)

- TO1: Imaging Interferometer (SDL)
Cryogenic Michelson Interferometer
Laser Metrology System
Calibration
- TO2: 128 x 128 IR FPA (BAE)
- TO2: Cryo-Cooler (TRW)
- TO3: Rad-Hard Analog to Digital Converters (Raytheon)
- TO3: Hi-speed Digital Filter/Co-adder (SDL)
- TO5: Autonomous Pointing & Control
Pointing Mirror Assembly (PMA) (SSG)
Star Tracker (T A&M)
512 x 512 Visible FPA (JPL)
- TO7: SiC Telescope (SSG)



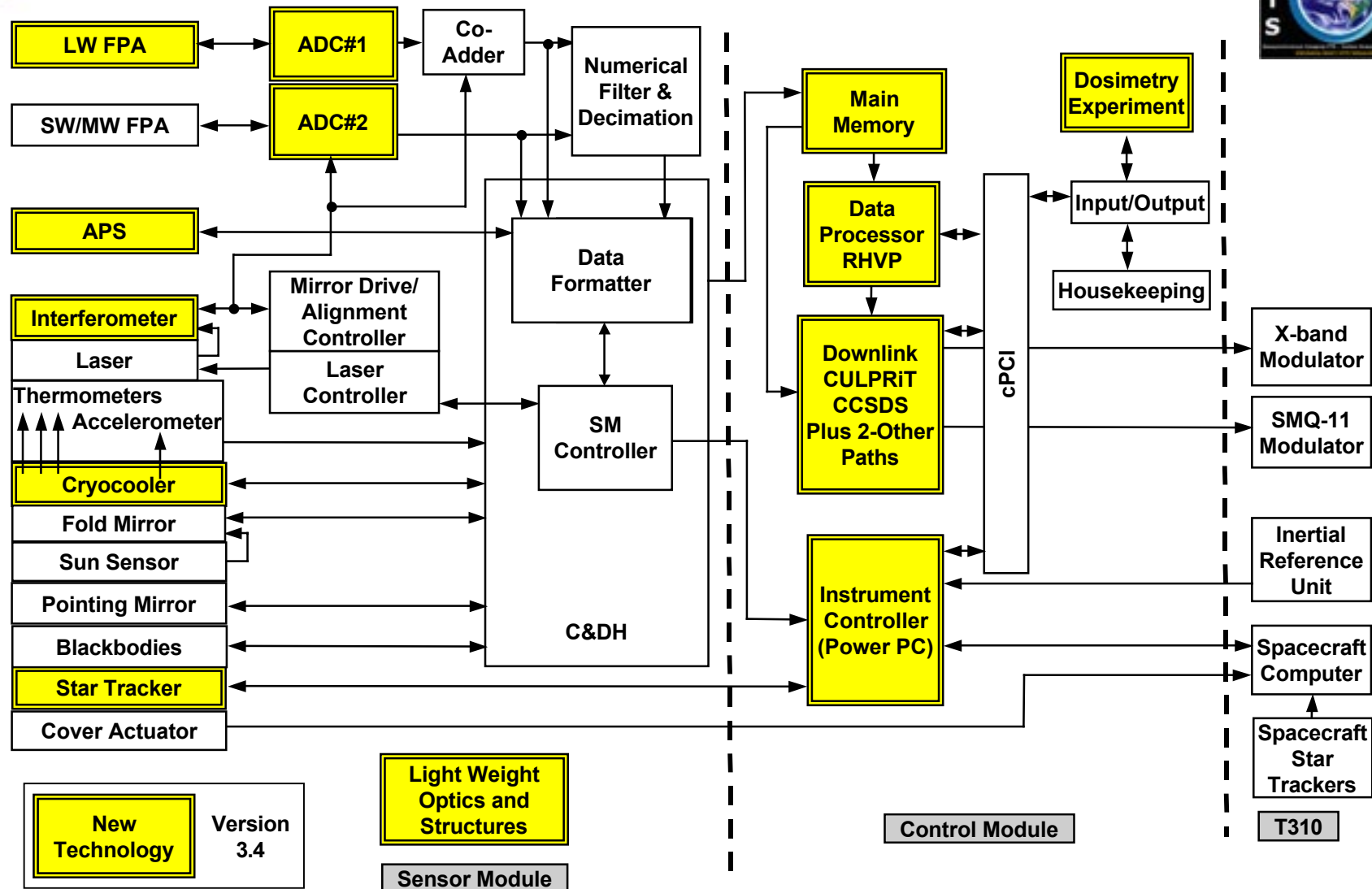
Control Module (CM)

- TO3: PowerPC Rad750 (BAE)
- TO3: Neo-Stack Memory (1GB) (ASRC/Irvine Sensors)
- TO4: Data Compression Experiment
Rad-Hard Vector Processor (Honeywell)
- TO6: CULPRIT CCSDS Downlink (UNM)
- TO6: Radiation Dosimetry Experiment
Shaped Shielding (LaRC)
Thermal Spray (Honeywell)
Composite Equipment Enclosure (COI)

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EO3 GIFTS Technologies for Validation



Validating technology at the system-level is challenging because some technologies are embedded and requires extensive operations and instrumentation planning for their validation.

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Technology Status Summary



	<u>TRL</u>	<u>Technical</u>			<u>Schedule</u>			<u>Cost</u>		
		Aug	Sep	Oct	Aug	Sep	Oct	Aug	Sep	Oct
Imaging Interferometer	5	G	G	G	G	G	G	G	G	G
Focal Plane Array	5	Y	Y	Y	G	Y	Y	G	G	G
Cryocooler	5	Y	Y	Y	Y	Y	Y	G	G	G
Rad-Hard Vector Processor	5	G	G	G	G	G	G	G	G	G
A/D Converter	5	G	G	G	G	G	G	G	G	G
Power PC	5	G	G	G	G	G	G	G	G	G
CULPRiT	4-5	G	Y	Y	G	G	G	G	G	G
Composite Optics	5	G	G	G	G	G	G	G	G	G
Stacked Memory	5	G	G	G	G	G	G	G	G	G
Radiation Protection	5	G	G	G	G	G	G	G	G	G
Advanced Pixel Sensor	5	G	G	G	G	G	G	G	G	G
Star Tracker	4	G	G	G	G	G	G	G	G	G

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Technology Status Summary



- GIFTS Technology Readiness:
 - 10 out of 12 technologies are currently at TRL 5
 - CULPRiT will reach TRL 5 by 3/02
 - Star tracker will reach TRL 5 by 6/02
 - All technologies will reach TRL 6 after completing instrument testing
- Guideline to deliver technologies (to SM & CM):
 - Approx \$10M actual expenditures to date
 - Approx \$43M estimated at completion



Mission Description

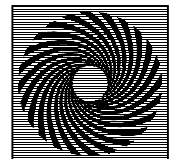


- GIFTS-IOMI is mission being conducted jointly by NASA, the Department of Navy (DON) Office of Naval Research (ONR), and NOAA.
- NASA Earth Observing-3 (EO-3)
 - EO-3, a mission in NASA's New Millennium Program (NMP) validates breakthrough technologies, and next generation measurement concepts
 - The GIFTS instrument provides unprecedented spatial and temporal weather soundings
- Indian Ocean METOC Imager (IOMI)
 - Uses the GIFTS instrument for a demonstration of operational utility.
 - Provides meteorological products directly to shipboard and shore assets in and around the Indian Ocean
- Two operational phases (EO-3 & IOMI)
- NOAA will support EO-3 Phase with a dedicated X-Band ground station.
- NOAA will enhance the utility of GIFTS through a NOAA-funded and executed Product Assessment Program.



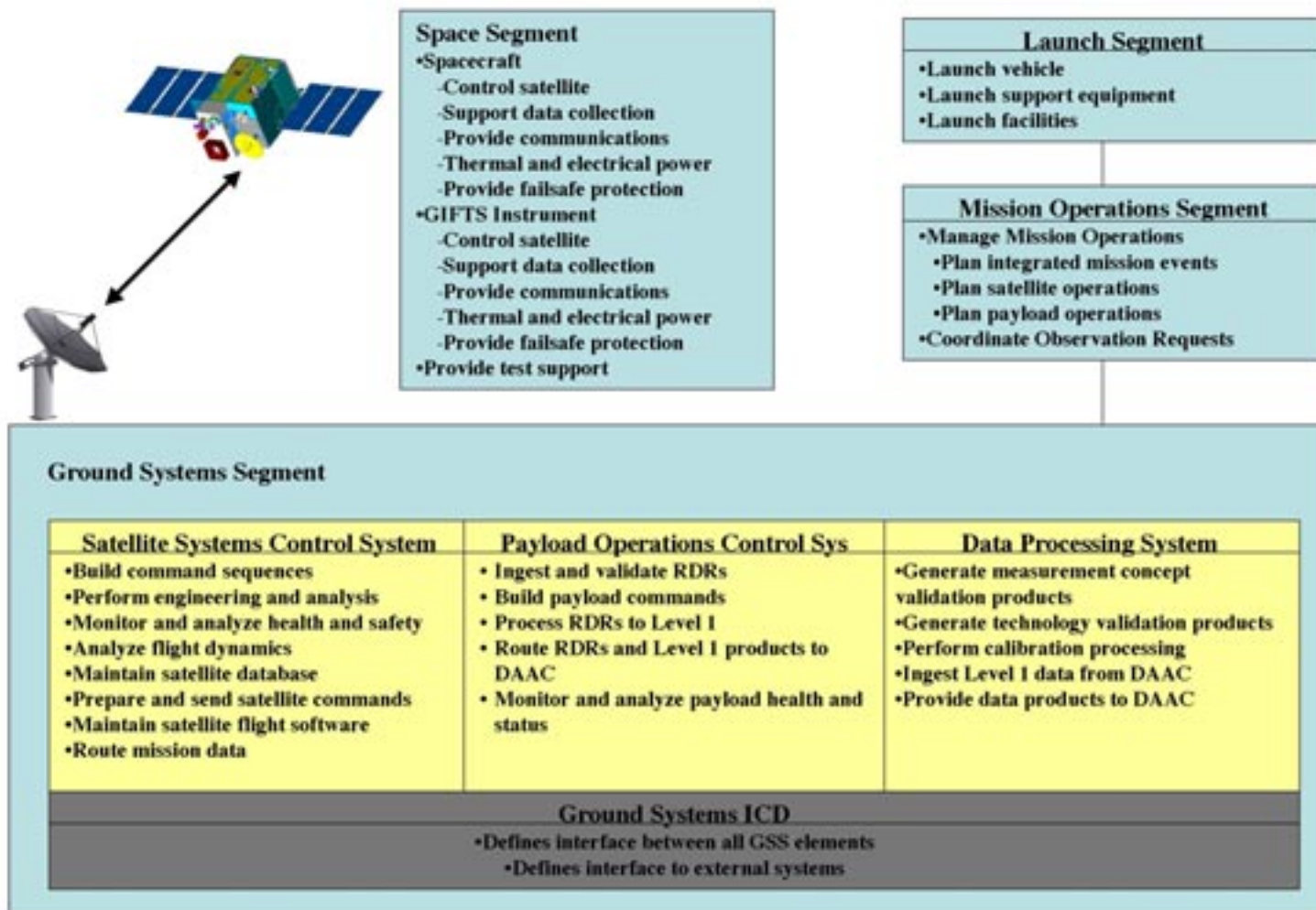
GIFTS/IOMI Proposed Operational Concept

GIFTS/IOMI Enables Improved Global Weather Prediction





Mission Architecture

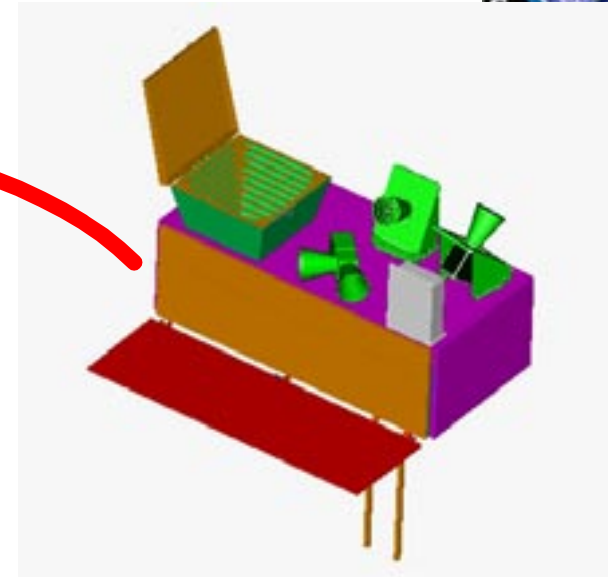
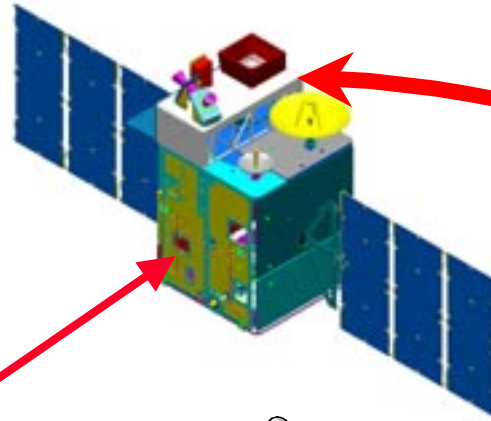




Space & Launch Segment Elements



Spacecraft TRW T-310



Control Module (CM)

PowerPC (Rad750)
Downlink I/O
Vector processor
Memory and SM I/O
Radiation Dosimetry board (1)

Delta-4 Launch Vehicle

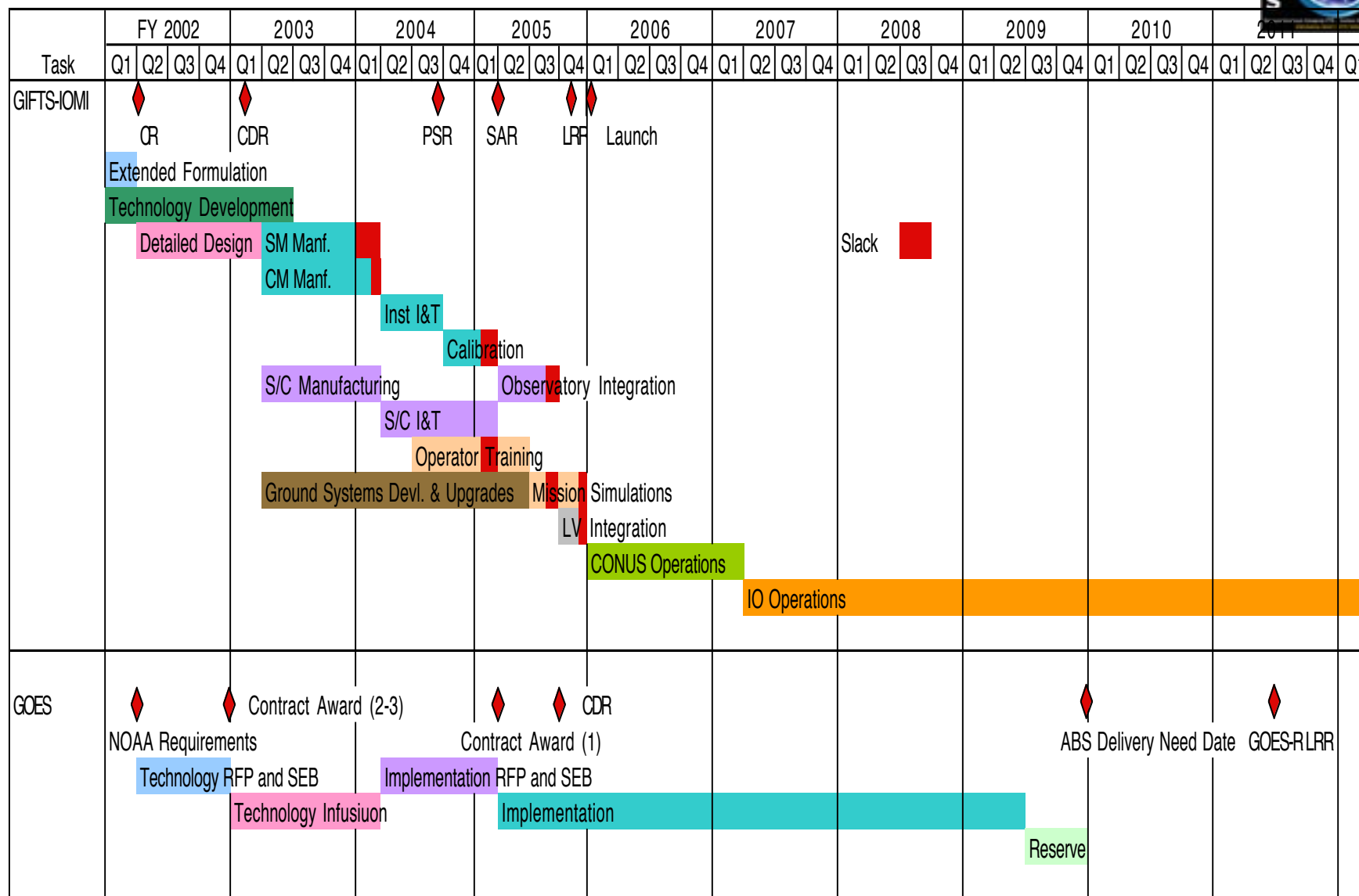


Sensor Module (SM)

Imaging Interferometer
128 x 128 IR FPA (2)
512 x 512 Visible FPAs
7 selectable resolutions (36-0.6 cm⁻¹)
Cryo-Cooler
Closed-loop Point and Control
Gimbal, Star Tracker, Gyro
Calibration Sources
Metrology System



Top Level Mission Schedule



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Current Mission Status



- System Requirements Review completed July 18, 2000.
- Preliminary Design Review completed March 20-22, 2001.
 - 26 Actions assigned
- 2 Independent cost estimates performed
 - Aerospace Corp., NASA Independent Program Assessment Office
 - Both validated GIFTS-IOMI project estimates
- Project directed to focus on technology maturation and instrument risk reduction activities while awaiting confirmation
 - NASA provided \$7M to support these activities
 - Mission development activities minimized
 - Comprehensive Technology Validation Planning Meeting conducted 10/24
- NOAA has adopted the GIFTS baseline for the Advanced Baseline Sounder (next generation GOES sounder)
- ATP into implementation is expected in Jan. 2002.